TEST - 2015

AT		COURSE	DAY: SUNDAY
AL	AE	RONAUTICAL	TIME: 10.00 A.M. TO 1.00 P.M.
MAXIMUM N	MARKS	TOTAL DURATION	MAXIMUM TIME FOR ANSWERING

	MENT				QUESTION BOO	KLET DETAILS
DI	PLOMA	CET	NUM	BER	VERSION CODE	SERIAL NUMBER
	1-3				A – 2	180074

180 MINUTES

DOs:

- 1. Check whether the Diploma CET No. has been entered and shaded in the respective circles on the OMR answer sheet.
- 2. This Question Booklet is issued to you by the invigilator after the 2nd Bell i.e., after 09.50 a.m.

200 MINUTES

- 3. The Serial Number of this question booklet should be entered on the OMR answer sheet and the respective circles should also be shaded completely.
- 4. The Version Code of this question booklet should be entered on the OMR answer sheet and the respective circles should also be shaded completely.
- 5. compulsorily sign at the bottom portion of the OMR answer sheet in the space provided.

DON'Ts:

- 1. THE TIMING AND MARKS PRINTED ON THE OMR ANSWER SHEET SHOULD NOT BE DAMAGED / MUTILATED / SPOILED.
- 2. The 3rd Bell rings at 10.00 a.m., till then;

180

- Do not remove the paper seal of this question booklet.
- Do not look inside this question booklet.
- Do not start answering on the OMR answer sheet.

IMPORTANT INSTRUCTIONS TO CANDIDATES

- 1. This question booklet contains 180 (items) questions and each question will have one statement and four answers. (Four different options / responses.)
- 2. After the 3rd Bell is rung at 10.00 a.m. remove the paper seal of this question booklet and check that this booklet does not have any unprinted or torn or missing pages or items etc., if so, get it replaced by acomplete test booklet. Read each item and start answering on the OMR answer sheet.
- 3. During the subsequent 180 minutes:
 - Read each question (item) carefully
 - Choose one correct answer from out of the four available responses (options / choices) given under each question / item. In case you feel that there is more than one correct response, mark the response which you consider the best. In any case, choose only one response for each item.
 - Completed darken / shade the relevant circle with a BLUE OR BLACK INK BALL POINT PEN against the
 question number on the OMR answer sheet.

Correct Method of shading the circle on the OMR answer sheet is as shown below:

- 4. Use the space provided on each page of the question booklet for Rough Work. Do not use the OMR answer sheet for the same
- 5. After the last Bells is rung at 1.00 p.m. stop marking on the OMR answer sheet and affix your left hand thumb impression on the OMR answer sheet as per the instructions.
- 6. Hand over the OMR ANSWER SHEET to the room invigilator as it is.
- 7. After separating the top sheet, the invigilator will return the bottom sheet replica (Candidate's copy) to you to carry home for self-evaluation.
- 8. Preserve the replica of the OMR answer sheet for a minimum period of ONE year.

PART - A APPLIED SCIENCE

1.	One	Pascal	is	equal	to
Τ.	OHC	1 ascai	10	cquai	-

3. The thrust on the bottom of the container having a base area of 20 m² filled with water to a height of 3 m is _____ (given
$$g = 10m/s^2$$
)

1.
$$6 \times 10^5 \text{ N}$$

2.
$$6 \times 10^4 \text{ N}$$

2.
$$6 \times 10^4 \text{ N}$$
 3. $6 \times 10^3 \text{ N}$ 4. $6 \times 10^2 \text{ N}$

4.
$$6 \times 10^2 \text{ N}$$

4.
$$-273^{\circ}$$
C

1.
$$\frac{1}{273}$$

2.
$$-\frac{1}{273}$$

1.
$$V = \sqrt{\frac{T}{m}}$$

2.
$$V = \sqrt{\frac{m}{T}}$$

3.
$$V=\sqrt{\frac{1}{T}}$$

2.
$$V=\sqrt{\frac{m}{T}}$$
 3. $V=\sqrt{\frac{1}{T}}$ 4. $V=\frac{\sqrt{m}}{T}$

			m Pi harbitata
9.	Absorption co-efficient of sound wave is given b		
	given medium E_{ow} is the energy absorbed by	open w	rindow.
	1. $a = \frac{E_m}{E_{ow}}$ 2. $a = \frac{E_{ow}}{E_m}$	3.	$a = E_m \times E_{ow} \qquad 4. a = E_m + E_{ow}$
10.	The rich quality of a musical note depends on		
	1. Fundamental frequency	2.	Loudness
	3. Larger number of over tones	4.	Pitch
11.	Waxing and waning are the characteristics of		
	1. Periodic motion 2. Oscillations	3.	Beats 4. Frequency
12.	Velocity of sound in air varies		
	1. Inversely as the square root of the densi	ty of t	he medium
	2. Directly as the square root of the density	y of the	e medium
	3. Directly as the density of medium		
	4. Inversely as the density of medium		
13.	The vibrations of a body of decreasing amplitu	ıde are	e called
	1. Undamped free vibrations	2.	Damped free vibrations
	3. Resonant vibrations	4.	Forced vibrations
14.	Another name for field emission is		
	1. Cold cathode emission	2.	Thermionic emission
	3. Photoelectric emission	4.	Secondary emission
15.	In case of photoelectric emission, the rate of	emissi	on of electron is
	1. Independent of frequency of radiation		

- 2. Dependent on frequency of radiation
- 3. Dependent on wavelength of incident radiation
- 4. Independent of intensity of radiation
- 16. Emission of radiation from radioactive element is
 - 1. Slow 2. Fast

3. Spontaneous

4. Very slow

17.		ne spectrum of scatte dent light are called	ered	light the lines co	rresp	onding to wavelength	greater than that of
	1.	Stokes lines			2.	Antistokes lines	
	3.	Fluorescent lines			4.	Incident lines	
18.	Reso	olving power of telesco	pe i	s given by			
	1.	$\frac{d}{1.22\lambda}$	2.	$\frac{1.22\lambda}{d}$	3.	$\frac{1.22d}{\lambda}$ 4.	$\frac{\lambda}{1.22d}$
19.	То о	bserve diffraction pat	tern	the obstacle shou	ld be		
	1.	Very big			2.	Dark	
	3.	Absent			4.	Comparable with the	e wavelength of light
20.		en double refraction o			ay ar	nd ordinary rays will l	have vibrations in the
	1.	Parallel	2.	Independent	3.	Perpendicular 4.	At 45°
21.	Max	well's electromagnetic	the	eory could explain	L		
	1.	Photo electric effect	;		2.	Interference of light	
	3.	Compton effect			4.	Black body radiation	n
22.	The	contrast between brig	ght a	and dark bands of	f an ir	nterference pattern is	
	1.	Low	2.	High	3.	No change 4.	. Gradually decreases
23.	A n	on-electrolyte solutior	ı is				
	1.	Sugar solution			2.	Salt solution	
	3.	Water			4.	Copper sulphate sol	ution
24.	In a	lkalies the concentra	tion	of OH^- ions is			

2.

Less than 10⁻⁷g ions / litre

More than 10⁷g ions / litre

More than 10⁻⁷g ions / litre

Equal to 10^{-7} g ions / litre

1.

3.

25.	An e	xample of derived uni	t is					
	1.	Meter	2.	Second	3.	Netwon	4.	Candela
26.	The p	prefix used for 10 ⁻¹⁵ is						
	1.	Femto	2.	Pico	3.	Peta	4.	Nano
27.	An e	xample of dimensionl	ess (constant is				
	1.	Strain	2.	Efficiency	3.	Force	4.	Pi
28.	A ma	ain scale is divided in t ofcm.	to h	alf mm and havin	g a V	Vernier containing	10	divisions has a least
	1.	0.05	2.	0.005	3.	0.02	4.	0.025
29.	Acco	rding to Newton's seco	ond 1	law of motion F = I	Kma.	The value of K is		
	1.	0.1	2.	0	3.	10	4.	1 -
30.	The v	velocity of a freely fall	ing l	body is maximum				
	1.	At the beginning			2.	Just before it tou	ches	ground
	3.	Exactly half way			4.	After it touches gr	roun	d -
31.	Wet	clothes are dried in w	ashi	ng machine by the	prop	perty of		
	1.	Inertia of rest			2.	Inertia of direction	n	
	3.	Inertia of motion			4.	Inertia of time		
32.	A for	ce of 1.2 x 10 ⁻² N acts ody is	s for	3 seconds on a boo	dy of	mass 0.04kg at res	st. T	he velocity gained by
	1.	0.9 m/s	2.	9 m/s	3.	0.09 m/s	4.	9.2 m/s
22	Λ.	grample of	4.24					
33.		xample of vector quar	_					
	1.	Volume	2.	Energy	3.	Density	4.	Force

34.	Handle of the door is fixed away from the end where it is fixed with hinges to							
	1.	Increase the moment of force	2.	Decrease the moment of force				
	3.	Keep the door firm	4.	Lock it easily				
35.	Resu	ltant of two equal forces perpendicular to e	ach o	ther acts at an angle to first force				
	1.	90° 2. 180°	3.	30° 4. 45°				
36.	The	resultant of two forces acting on a body can	not be					
	1.	Greater than first force						
	2.	Zero						
	3.	Lesser than first force						
	4.	Lesser than the difference between two for	rces					
37.	Towi	ng of a boat by two forces is an illustration (of					
	1.	Lami's theorem	2.	Law of triangle of forces				
	3.	Law of parallelogram of forces	4.	Law of polygon of forces				
38.	Shoo	ck absorber is an example for						
	1.	Compressive stress	2.	Tensile stress				
	3.	Shear stress	4.	Shear strain				
39.	Fact	or of safety of a structure is						
	1.	Within 2	2.	Equal to zero				
	3.	Vary between 5 and 10	4.	More than 10				
40.	In c	ase of liquids as the temperature increases	, the	viscosity of liquid decreases due to				
	1.	Increase in the rate of diffusion of gases						
	2.	Decrease in the rate of diffusion of gases						
	3.	Increase in the potential energy of mole	cules					
	4.	Increase in the kinetic energy of molecu	ıles					

PART - B

APPLIED MATHEMATICS

- 41. If $x \sin 30^\circ Sec 30^\circ \tan 30^\circ = \tan^2 60^\circ$, then the value of x is
- 2. $\frac{-22}{3}$ 3. $\frac{11}{6}$
- 4. $\frac{3}{22}$

- 42. The value of $sin 225^{\circ} + cos(-135^{\circ})$ is
 - $\sqrt{2}$
- 2. $-\sqrt{2}$
- 3. $\frac{1}{\sqrt{2}}$
- 4. $\frac{-1}{\sqrt{2}}$
- The simplified value of $\frac{\sin(180^{\circ}-A)\cot(90^{\circ}-A)\cos(360^{\circ}-A)}{\tan(180^{\circ}+A)\tan(90^{\circ}+A)\sin(-A)}$ is
 - sin A 1.
- 2. -sin A
- 4. cosec A

- The simplified value of $\frac{\sin 2A}{1+\cos 2A}$ is
 - 1. 2tan A
- 2. sin A
- cot A
- 4. tan A

- If $\tan A = \frac{3}{4}$ and $\tan B = \frac{1}{7}$, then the value of (A+B) is
 - 1. $\frac{\pi}{6}$

- $2, \frac{25}{23}$
- 3. $\frac{\pi}{4}$

- The value of $\cos 20^{\circ} + \cos 100^{\circ} + \cos 140^{\circ}$ is
 - 0 1.

- 2. cos 50°
- 4. sin 50°

47. The value of $\cos^{-1} \left[\tan 135^{\circ} \right]$ is

1. 0° 2. 180°

3. 45° 4. 90°

The centroid of the triangle formed by the vertices (-10, 6), (2, -2) and (2, 5) is

(-2, 3)

2. (2,3) 3. $\left(-3,\frac{9}{2}\right)$ 4. (-6,9)

A point (-4, 3) divides the line AB externally in the ratio of 1: 2. Given A(-1, -3) then the point B

(6, -3)1.

2. (-10, 15) 3. (2, 9) 4. (2, -9)

The area of triangle formed by the point, (3, -1), (2, 0) and (K, 4) is 10 Sq. Units, then the value of K

1. 12 2. 7

3. -22 4. 22

The slope of the line joining the points (-2, 3) and (4, -6) is

1.

2. $\frac{-3}{2}$ 3. $\frac{2}{3}$

The equation of straight line passing through (4, -1) and having equal intercepts is

x+y-1=01.

2. x+y-5=0

3. x+y-3=0 4. x+y+3=0

The equation of the line passing through (5, -2) and parallel to the line 3x + 2y + 7 = 0 is 53.

1. 3x + 2y - 11 = 0 2. 3x-2y+11=0

3x - 2y - 19 = 03.

2x-3y-16=0

- 54. The value of $\lim_{x \to -2} \frac{x+2}{x^5+32}$ is
 - 1. $\frac{1}{80}$
- 2. 80
- 3. $\frac{-1}{80}$
- 4. -80

- 55. The value of $\lim_{x \to 0} \frac{2x tan 3x}{\sin 2x + 3x^2}$ is
 - 1. $\frac{-1}{5}$
- 2. 0
- 3. $\frac{1}{2}$
- 4. $-\frac{1}{2}$

- 56. If $y = e^x \log x$, then $\frac{dy}{dx}$ at x = 1 is
 - 1. e^x

- 2. e
- 3. 1
- 4. 0

- 57. If $y = tan^{-1} \sqrt{\frac{1 + \cos x}{1 \cos x}}$, then $\frac{dy}{dx}$ is
 - 1. 2

- 2. -2
- 3. $\frac{-1}{2}$
- 4. $\frac{1}{2}$

- 58. If $\sqrt{x^3} + \sqrt{y^3} = \sqrt{a^3}$, then $\frac{dy}{dx}$ is
 - 1. $\sqrt{\frac{x}{y}}$
 - 2. $-\sqrt{\frac{x}{y}}$
- 3. $\sqrt{\frac{y}{x}}$
- 4. $-\sqrt{\frac{y}{x}}$

- 59. The second derivative of y = log(sec x tan x) is
 - 1. $-\sec x \tan x$
- $2. \quad sec \ x \ tan \ x \qquad \qquad 3.$
- 3. -sec x
- 4. sec x

- Water flows into the cylindrical tank of radius 7mt at the rate of 294 cubic mt/sec, then the rate of 60. height of water rising in the tank is
 - $\frac{\pi}{6}$ mt / sec

2. $\frac{6}{\pi}$ mt / sec

14406 mt / sec 3.

- 4. $\frac{21}{\pi}$ mt / sec
- The maximum value of the function $y = x + \frac{1}{x}$ is
 - 1. 0

- 2. 2
- 3. 1

- The value of $\int tan^2x \ dx$ is
 - 1.
- tan x-x+c 2. x-tan x+c 3. $(sec^2x)^2+c$ 4. -cot x-x+c

- 63. The value of $\int \frac{\cos x}{1+\sin x} dx$ is
 - 1. $log(sec^2x + sec x tan x) + c$
- $2. \quad \log(\sin x) + c$

3. log(1+sin x)+c

4. $\frac{(1+\sin x)^2}{2}+c$

- 64. $\int \sin^2 x \sin 2x \, dx$ is
 - 1. $\frac{\sin^2 x}{2} + c$
- $2. \quad \frac{\sin^4 x}{2} + c$
- 4. $\frac{-\sin^4 x}{2} + c$

65.
$$\int_{-1}^{1} (2x+1)(5-x) dx$$
 is

10

- 2. $\frac{26}{3}$
- 3. $\frac{-26}{3}$
- 4. $\frac{11}{3}$

66.
$$\int_{0}^{\frac{\pi}{4}} tan^{2}x \ sec^{2}x \ dx$$
 is

- 2. $\frac{4}{3}$
- 3. $\frac{1}{2}$
- 4. $\frac{-1}{3}$
- The RMS value of $y^2 = x^2 2x$ over the interval [1, 3] is
 - 1. $\sqrt{\frac{5}{3}}$
- 2. $\sqrt{\frac{2}{3}}$
- 4. $\frac{1}{\sqrt{3}}$
- The differential equation of $y^3 = 5 ax$ by eliminating arbitrary constant a is
 - 1. $\frac{dy}{dr} \frac{y}{3r} = 0$

 $2. \quad \frac{dy}{dx} + \frac{y}{3x} = 0$

3. $\frac{dy}{dr} - \frac{3y}{r} = 0$

- 4. $\frac{dy}{dr} \frac{5y}{3r} = 0$
- The integrating factor of the differential equation $x \frac{dy}{dx} (1-x)y = x^3$ is
 - 1. $\frac{e^x}{r}$
- 2. xe^x
- 3. $e^{\frac{x^2-2x}{2}}$ 4. $e^{\frac{2x-x^2}{2}}$

- 70. If $\begin{vmatrix} 2x+1 & -5x \\ 1 & 3 \end{vmatrix} = 0$, then x is
 - 1. $\frac{3}{11}$

- 2. $\frac{-3}{11}$ 3. $\frac{11}{3}$ 4. $-\frac{11}{3}$
- 71. For the simultaneous linear equations 2x + y + z = 1, x + y + 2z = 0 and 3x + 2y z = 2, the value of Δx is
 - 1.

- 2. -11

- 72. If $A = \begin{bmatrix} 2 & 3 \\ 5 & 4 \end{bmatrix}$, $B = \begin{bmatrix} -1 & 7 \\ -4 & 1 \end{bmatrix}$ then $(A+B)^T$ is
- 1. $\begin{bmatrix} 1 & 1 \\ 10 & 5 \end{bmatrix}$ 2. $\begin{bmatrix} 1 & 10 \\ 1 & 5 \end{bmatrix}$ 3. $\begin{bmatrix} -1 & 10 \\ -1 & 5 \end{bmatrix}$ 4. $\begin{bmatrix} -1 & -1 \\ 10 & 5 \end{bmatrix}$

- 73. If $A = \begin{bmatrix} 1 & -3 \\ -5 & 7 \end{bmatrix}$, then adj A is
 - 1. $\begin{bmatrix} 1 & -5 \\ -3 & 7 \end{bmatrix}$ 2. $\begin{bmatrix} 7 & -5 \\ -3 & 1 \end{bmatrix}$ 3. $\begin{bmatrix} -1 & -5 \\ -3 & -7 \end{bmatrix}$ 4. $\begin{bmatrix} 7 & 3 \\ 5 & 1 \end{bmatrix}$

- 74. The cofactor of O in $A = \begin{bmatrix} 3 & -2 & 5 \\ 1 & 6 & 0 \\ 2 & 7 & -4 \end{bmatrix}$ is
 - -25

4. 0

- 75. If $(\sqrt{3}+1)^3 = 10+6\sqrt{3}$, then the value of $(\sqrt{3}+1)^3-(\sqrt{3}-1)^3$ is
 - 1. 12√3
- 2. 0
- 20
- 4. $20 + \sqrt{3}$

- The middle term in the expansion of $\left(x^3 + \frac{1}{x^2}\right)^6$
 - $10 x^3$ 1.
- 2. $20 x^3$ 3. $\frac{20}{x^3}$
- 4. 20

- 77. If $\vec{a} = i + 3j 2k$ and $\vec{b} = 2i j + 3k$, then $\vec{a} \cdot \vec{b}$ is

- 2. 11

- The work done by the force 2i j + 6k when it displaces the particle from (5, 3, -2) to (7, -4, 8) is
 - 1. 72

- 2. 48

- The sine of the angle between the vectors $\vec{a} = i + j + k$ and $\vec{b} = 2i 3j 4k$ is
 - 1. $\sqrt{\frac{62}{87}}$
- 2. $\sqrt{\frac{87}{62}}$ 3. $\frac{-5}{\sqrt{87}}$
- 4. $\sqrt{\frac{10}{63}}$
- 80. If $\cos \theta = \frac{5}{13}$ and θ is acute angle, then the value of $3\cos \theta 2\sin \theta$ is
 - $1. \qquad \frac{9}{13}$

- 2. 3
- 4. -3

PART - C

AERONAUTICAL ENGINEERING

- 81. COANDA effect is made use of in modern aircraft
 - 1. To get additional speed
 - 2. To get additional lift during landing, take-off
 - 3. During vertical climb mode for a fighter
 - 4. For getting zero drag condition in cruising
- 82. Coefficient of Lift C_L for an aircraft
 - 1. Keeps on increasing with speed
 - 2. Becomes steady or constant in supersonic regime of operation
 - 3. Is always constant for a type of aircraft at all speeds
 - 4. Keeps on varying in a linear manner
- 83. In sky wave propogation Electromagnetic waves
 - 1. Always pass through the ionosphere for better communication
 - 2. Reflect, refract and always come back to place of origin
 - 3. Reflect, refract and always come back to a distance place
 - 4. Reflect, does not retract and reach to nearby place
- 84. In LOS communication, the Range is dependent on
 - 1. Distance to be sent
 - 2. Power of transmission
 - 3. Obstacles on the way
 - 4. All of these factors

- 85. In modern aircraft, the most preferred type of communication used are
 - 1. HF because the range is very big
 - 2. VHF only due to clear communication
 - 3. VHF and UHF both
 - 4. VHF, UHF and SHF as all range are covered
- 86. The public address system in an aircraft
 - 1. Is used to take the addresses of all passengers
 - 2. Is a critical equipment and is not switched off in power failures
 - 3. Is a non-critical equipment which can be switched off during power failure
 - 4. Is used only to give speech by pilots and airhostesses
- 87. An ADF equipment in an aircraft gives
 - 1. Automatic data function for pilot
 - 2. Gives directional guidance or data in horizontal plane
 - 3. Gives directional data in vertical and horizontal planes
 - 4. Gives kilometer to go information to pilots
- 88. An ILS equipment gives the pilot information on
 - 1. All the instrument in the aircraft
 - 2. All the altitudes of aircraft during cruise
 - 3. Deviation from approach elevation angle and centre of runway
 - 4. Deviation from standard speed of approach
- 89. MLS equipment is better compared to ILS because
 - 1. It is much more economical compared to ILS
 - 2. It is much more accurate and gives better approach option for pilots
 - 3. ILS is already obsolete, and is not supported in many countries
 - 4. ILS equipment is too heavy and consumes a lot of power

90.	GPS	navigation	makes	use	of
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- 1. 48 satellites in GEO orbit
- 2. 24 satellites in MEO orbit
- 3. 21 satellites in LEO orbit
- 4. 23 satellites in POLAR orbit

91. In an intertial Navigation System (INS), the accuracy of data

- 1. Degrades with altitude
- 2. Degrades with time
- 3. Degrades with time and distance from starting point
- 4. Improves with time and distance from starting point

92. Bonding in aircraft electrical wiring system is done

- 1. To have a correct fixing off full wiring
- 2. To have a common continuity between all portions or parts of aircraft
- 3. To have a correct impedance matching
- 4. To have least resistance in all power circuits

93. The squirrel cage induction motor is very popular and extensively used because

- 1. The construction is very easy
- 2. The cost is very economical
- 3. Wear and tear of parts is very low
- 4. Brushes can be charged easily and quickly

94. AC alternators are working in aircraft on

1. 25V AC, 50 C/S

2. 25V DC, 400 C/S

3. 115V AC, 400 C/S

4. 115V DC, 400 C/S

95.	400C/S frequency is used in aircraft AC circuit because							
	1.	50 C/S frequency is not economical						
	2.	It reduces the heating and induction losse	es					
	3.	It is available in all countries						
	4.	It reduces the weight of all AC machines						
96.	A car	rbon pile regulator is very popular in aircra	ft reg	gulating circuit because				
	1.	It reduces the total cost	2.	It has no parts with move				
	3.	It has no contact pitting problem	4.	It is developed for only aircrafts				
97.	Aircr	aft lighting system operates on						
	1.	AC	2.	DC				
	3.	Both AC and DC	4.	Switch mode power supply system				
98.	In fo	ur-stroke engine one power stroke is delive	red b	y revolution of crankshaft				
	1.	1 2. 4	3.	8 4. 2				
99.	In ar	a aircraft piston engine propeller speed is co	ontro	lled by				
	1.	Reduction in pressure	2.	Reduction gear box				
	3.	Accelerator	4.	Brake system				
100.		of an IC engine converts rotary to li	inear	motion				
	1.	Crank shaft	2.	Main shaft				
	3.	Propeller shaft	4.	Camshaft				
101.	Engi	ne frictional power is defined as						
	1.	IP + BP 2. IP / BP	3.	IP-BP 4. IP X BP				
102.	Engi	ne oil is usually cooled by						
	1.	Ramair	2.	Bleed air				
	3.	Fuel / oil cooler	4.	Hot exhaust gas				

103.	The a	approximate per	centage o	f the mass ai	rflow take	n in by the flam	e tube	snout is	
	1.	82 %	2.	8 %	3.	50 %	4.	18 %	
104.	Thru	st reversal on a	High Byp	oass Engine is	s achieved	by			
	1.	Bloker Door			2.	Clamshell			
	3.	Bucket type do	or		4.	Values			
105.	Nor	mal gas turbine	engine's	exhaust duct	is				
	1.	Divergent			2.	Convergent /	Diverg	gent	
	3.	Convergent			4.	Parallel			
106.	The	highest turbine	bearing	temperature	take place	•			
	1.	All the time	2.	At start up	3.	At shutdown	4.	At cruising	
107.	Kerc	sene will burn e	effectively	at an air to f	fuel ratio o	of			
	1.	150 : 1	2.	15:1	3.	45:1	4.	75 : 1	
108.	The	function of the s	station va	ne assembly a	at the disc	harge end of a ty	ypical	axial flow compres	ssor
	1.	To increase air	r swirling	motion into	combustic	on			
	2.	To direct flow of	of gases in	nto combusto	r				
	3.	To strengthen	airflow to	o eliminate ti	urbulence				
	4.	To increase tu	ırbulence						
109.	Тур	e of Oil system u	sually for	und on Turbi	ne engines	3			
	1.	Dry sump, Dip	and Spla	sh	2.	Dry sump, Pre	ssure	and Spray	
	3.	Wet sump, Spi	ray and S	plash	4.	Wet sump Dip	and S	plash	
110.		Engine	is used	for powering	helicopter				
	1. T	urbo fan	2.	Turbo jet	3.1	ľurbo prop	4.	. Turbo shaft	
111.	Con	nbustion chambe	er in whi	ch individual	liner and	d a common airc	asing	is used is called	

112.	Gas	turbine engine opera	tes c	on				
	1.	Carnot Cycle	2.	Otto Cycle	3.	Brayton Cycle	4.	Duel Cycle
113.		al examination and m	anu	al checks to deter	mine	the condition of a	n air	craft or components
	1.	Inspection	2.	Quality control	3.	SQC	4.	Maintenance
114.	Bird	strike is sometimes	calle	d as				
	1.	Aviation ingestion	2.	Bird hit	3.	BASH	4.	All of the above
115.	The	pneumatic leak check	k is	designed to check	the	cabin for		
	1.	Air tightness	2.	Pressure	3.	Temperature	4.	Volume
116.		causes uneven	trea	ad wear, loss brak	ing ef	ficiency, abnorma	l tyre	e growth, bulging out
	1.	Under inflation			2.	Over inflation		
	3.	Normal inflation			4.	No inflation		
117.		ring the aircraft aga			emen	t and protecting	agai	nst adverse weather
	1.	Mooring	2.	Tawing	3.	Taxing	4.	Flying
118.	Fire	caused by petrol, die	sel,	thinner, oil is clas	ssified	l as		
	1.	Class D	2.	Class A	3.	Class B	4.	Class C
119.	The	record in which all d	ata (concerning the air	craft	is recorded is cal	led _	
	1.	Log book			2.	Maintenance M	anua	d
	3.	IPC			4.	Handbook		
120.	The	cabin air conditionin	g an	d pressurisation s	syster	n supplies conditi	oned	air for
	1.	Heating			2.	Cooling		
	3.	Heating and cooling	3		4.	None of the above	ve	

121.	Life c	ycle refers to the period		
	1.	From design to launch	2.	From first launch to final withdrawal
	3.	From manufacturing to launch	4.	From manufacturing to inpsection
122.	Availa	ability is		
	1.	Item is in ready to use state	2.	Breakdown
	3.	Correction before use	4.	Failure in use
123.	MDT	means		
	1.	Mean Doubt Time	2.	Mean Delay Time
	3.	Mean Decay Time	4.	Mean Down Time
124.		teral assurance that certain items or servivent of failure is	ices a	as factually stated and provides remedy in
	1.	Replacement 2. Repair	3.	Warranty 4. Maintenance
125.	If a fa	ailure of one component cannot cause relate	ed fail	ure in other components is called
	1.	Secondary failure	2.	Metallurgical failure
	3.	Independent failure	4.	Creep failure
126.	Main			disease posing serious health risk is called
	1.	Prophylactic maintenance	2.	Conditional maintenance
	3.	Corrective maintenance	4.	None of the above
197	EDB	means		
121.	1.	Free Data Recorder	2.	Flow Data Recorder
	3.		2. 4.	Federal Data Recorder
	J.	Flight Data Recorder	٦.	reactal Data Recorder
128.	comp			nine whether quality activities and result se arrangement are implemented effectively
	1.	Audit 2. Quality circle	3.	TQM 4. ISO

129.	Failure which is both gradual and partial is known as									
	1.	Early failure			2.	Catastrophic fail	ure			
	3.	Degradation failure			4.	Partial failure				
130.		ne parts of the entire matically in		rcraft is maintair —	ned w	vith reference nu	reference number, part number and			
	1.	Catalogue			2.	Illustrated parts	cata	logue		
	3.	Log book			4.	Maintenance Ma	nua			
131.	Movi	ng the control column	ı for	ward, the effect is						
	1.	Pitch up		Pitch down	3.	Roll port	4.	Roll starboard		
132.	Move	ment of aircraft above	e ve	rtical axis is						
	1.	Rolling	2.	Pitching	3.	Yawing	4.	Diving		
133.	Bell	crank is used in								
	1.	Cable pulley system			2.	Push pull rod sys	stem			
	3.	Chain sproket syste	m		4.	None of the abov	е			
104	/INIa.a.	and the second subjects of a		the bridgerylic flyid	und	or processes is				
154.		component which sto			3.	Power pump	4	Accumulator		
	1.	Reservoir	2.	Selector	٥.	rower pump	٦.	Accumulator		
135.	Conv	entional undercarria	ge l	ias						
	1.	None wheel			2.	Tail wheel				
	3.	Fair undercarriage			4.	None of the abov	re			
136.	Turb	o-coolers are used in								
	1.	Air cycle cooling sys	stem		2.	Freon gas refrig	erati	on		
	3.	Both are correct			4.	Both are incorre	ct			
137	Hydr	aulic system fluid ch	0086	n should have						
107.	1.	High flash and fire			2.	Low flash and fir	e po	int		
	3.	Medium flash and f			4.	All of the above				

138.	Mine	ral based hydraulic flu	iid (colour is				
	1.	Blue	2.	Green	3.	Yellow	4.	Red
139.	Varia	able restrictor is used	in					
	1.	Wiper system			2.	Aircraft doors		
	3.	Both are correct			4.	Both are incorrec	et	
140	Stati	c pressure is utilised	hve					
140.			Dу 2.	Altimeter	3.	VSI	4	All of the above
	1.	ASI	4.	Attilleter	J.	V 51	т.	An of the above
141.	Mark	the correct statemer	nt					
	1.	Air speed is proportion		to P _n -P _n				
	2.	Altitude is proportion						
	3.	Air speed is inversely)			
	4.	None of the above			0			
142.	Squa	are Law compensation	is	achieved by				
	1.	Dimensioning of tra	nsn	nitting media	2.	Tuning spring		
	3.	Profiled cams			4.	All of the above		
143.		ersonic speed means						
	1.	Less than 1 M			2.	Less than 2 M		
	3.	Less than 3 M			4.	More than 5 M		
144	Stan	ndard compass swingir	10 C	ealibration is done	at			
177.	1.	Four heading				Twelve heading		4 Six heading
	1.	roul licauling	4.	Eight heading	Э.	Twelve Heading		i. Ola licaulis
145.	Airc	raft component log bo ond economical repair	oks	are to be preserve	ed for	a period of		_ years after damage
	1.	0.5 years	2.	1.0 year	3.	1.5 years	4.	2.0 years

146.	Flight manual section - 3 contains								
	1.	General information			2.	Limitation impos	ed		
	3.	Emergency procedur	es		4.	None of the above	е		
147.	India	n registered aircraft	has	nationality mark	starti	ing with letter			
	1.	VT	2.	IN	3.	Н	4.	ВН	
148.	Statically unstable structure is given by								
	1.	M = 2J - 3			3.	M < 2J - 3	4.	M > 2J - 3	
149.	Whic	h is not a part of Trus	ss ty	pe of fuselage con	istruc	tion ?			
	1.	Frame			2.	Longeron			
	3.	Diagonal member			4.	Bulkhead			
150	In a	reaton miles wing the	ato1	l first seemen at					
150.		rectangular wing the				20.131		W751-4	
	1.	Tip	2.	Root	3.	Middle	4,	Winglet	
151.		erally, wing with ondition of flying		configuration	provi	des longitudinal a	nd la	ateral stability during	
	1.	Mid wing			2.	Dihedral wing			
	3.	Low wing			4.	Anhedral wing			
152.	Identify the lateral member placed at regular interval from root to tip and from leading edge to trailing edge of wing								
	1.	Spar	2.	Ribs	3.	Stringer	4.	Anti-drag wire	
153.	In a	n T' Beam, the maxim	um	shear stress is ca	arried	by			
	1.	Web			2.	Flange			
	3.	Between web and fla	ange		4.	Neutral axis			
0.00	Space For Rough Work								

154.	Which control surface is located at the trailing edge of the horizontal stabilizer?							
	1.	Rudder	2.	Aileron	3.	Elevator	4.	Spoilers
155.	Wing	lets are fitted at the t	ip se	ection of wing to re	educe	:		
	1.	Skin friction drag			2.	Profile drag		
	3.	Induced drag			4.	Wave drag		
156	Air s	afety organisation pre	ven	ts failure through	regu	lation.	aı	nd training
100.	1.	Design		is amure timengar	2.	Education		
	3.	Modern			4.	Material		
157.	What	t is the full form of DG	CA 1	?				
	1.	Direct General of Ci	vil A	viation				
	2.	Director General of	Civil	Aviation				
	3.	Director General of	Com	mercial Aviation				
	4.	Direct General of Co	mm	ercial Aviation				
150	In th	e below type of ATF v	whio	h has the highest	Floor	Point 2		
130.	1.	Jet A-1		Jet A	3.	Jet B	4	JP-5
	1.	Jet A-1	۷.	Jet A	٥.	Jet B	т.	01-0
159.	Norn	nally, newly filled tyre	is a	llowed to stretch i	for a	minimum period of	í	hours
	1.	10	2.	5	3.	12	4.	24
160.	In ca	ase of low r.p.m engine	e fai	lure the blade will				1
	1.	In direction			2.	Opposite direction		
	3.	In perpendicular dir	ectio	on	4.	None of the above	•	
161.	Slop	oping of instrument panel to degree to minimise parallex error						
	1.	50		15	3.	10		25
162.		type of display	pre	sents data withou	t req	uiring users to loo	k av	vay from their usua
	1.	HDD	2.	Radial	3.	HUD	4.	Vertical

163.	is used to have a non-continuous indication of pressure							
	1.	Pressure gauge			2.	Pressure switch		
	3.	Tachometer			4.	Bourdon tube		
164.	In W	heat-Stone Bridge, the	e ba	lanced condition r	efers	to		4
	1.	High current			2.	Low current		
	3.	Zero current			4.	Opposite current		
165.	Meas	urement of RPM in e	ngin	e using light is ca	alled a	as		
	1.	Tachogenerator			2.	Stroboscope		
	3.	Synchronoscope			4.	Tacho indicator		
166.	The j	property of material to	э гез	sist penetration is	calle	d.		
	1.	Brittleness			2.	Hardness		
	3.	Ductility			4.	Malieability		
167.	High	carbon steel consists	of _	% of carbo	n			
	1.	1.2 to 1.5	2.	0.5 to 1	3.	0.3 to 0.5	4.	0 to 0.2
168.	The	second most importar	nt co	onstituent of stain	less	steel is		
	1.	Tungsten	2.	Carbon	3.	Chromium	4.	Titanium
169.	Whic	h is not the advanta	ige (of composite mate	rial?			
	1.	High strength			2.	Low cost		
	3.	Does not corrode			4.	Less weight		
150	****							
170.		h among the followin	g is	surface inspection				
	1.	DYE penetration			2.	Eddy current		
	3.	Ultrasonic			4.	Thermal		

171. Atmospheric Air contains

- 1. $N_2 78\% O_2 21\%$ Inert gases -1%
- 2. $N_2 80\% O_2 19\%$ Other gases -1%
- 3. $N_2 75\% O_2 20\% CO_2 2\% Other gases 3\%$
- 4. $N_2 80\% O_2 20\%$

172. As per ISA standard temperature, pressure, relative humidity at MSL are

- 1. $T = 15^{\circ}C$, P=1018.25mb Relative humidity =100%
- 2. T=15°C, P=1018.25mb Relative humidity = 98%
- 3. $T=15^{\circ}C$, P=1013.25mb, Relative humidity = 100%
- 4. T=17°C, P=1030.25mb, Relative humidity =100%

173. Lapse rate for atmospheric temperature and pressure are

- 1. 3°C per 100 ft climb, 1mb per 35 ft climb
- 2. 1.98°C per 1000ft climb, 1mb per 30 ft climb
- 3. 1.98°C per 100ft climb, 1mb per 100 ft climb
- 4. 1.98°C per 10ft climb, 1mb per 50 ft climb

174. Secondary control surface are used

- 1. Whenever pilot feels loss of lift
- 2. Only during bad weather
- 3. Only during cruising
- 4. Only during take-off and landing

175. Flaps are used by the pilot to

- 1. Control high speed manoeuvres
- 2. Control angle of attack during take-off, landing
- 3. Obtain lift during reduced speeds on landing and take-off
- 4. Get steady, cruise speed control for auto pilot mode

176. Seats are used for

- 1. Getting additional control for auto pilot
- 2. Getting additional lift during landing, take-off at reduced speeds
- 3. Getting instant control of aircraft altitudes
- 4. Getting a very smooth cruise control

177. Slots are

- 1. Part of flaps
- 2. Additional devices for controlling boundary layer separation
- 3. Devices used for additional drag
- 4. Device used for getting additional speed in climb and dive

178. Winglets are used in aircraft wings

- 1. To reduce the wing tip vortices to a large extent
- 2. To reduce the aspect ratio of the wing
- 3. To reduce the area of the wing
- 4. To reduce the drag on leading edge

179. NACA is the short form for

- 1. National Aircraft Committee for Aviation
- 2. National Aeronautics Committee of America
- 3. National Aviation Council for Aeronautics
- 4. National Authority for Counselling Aviation

180. The wake behind a big aircraft extends to

- 1. Upto 2-3 kms behind the aircraft
- 2. Dies down within 100 metres
- 3. Upto 15 kms behind the aircraft
- 4. Upto 500 metres depending on weather conditions